

April 26, 2018

Dylan Vandemark CellBlock FCS, LLC 234 Northeast Road Unit #5 Standish, Maine 04084

Dear Mr. Vandemark:

Enclosed are the results of the 12 Second Vertical Burn Test for CellBlock FCS, LLC Standish, ME. Test specimen description as follows:

TEST NUMBER(S)	TEST SAMPLE PART NUMBER(S)
V18 1839	3A
V18 1840	PGL2000
V18 1841	SGL1700
V18 1842	ASM250

The tests were conducted in accordance with Title 14 CFR, PART 25—Subpart D, § 25.853 (a) compartment interiors, [Amdt. 25-116, Nov. 26, 2004] Appendix F to Part 25, Part I (a)(1)(ii) [Amdt. 25-142, Eff. 4/18/2016] and FAA Approved "AccuFleet Testing Procedures" dated May 1995. The test operator was Elizabeth Huff and witness present was Shawn K. King, DERT Consultant.

If you have any questions regarding this test, please call me at 281-999-8800 extension 1111 or 1134. Sincerely,

ACCUFLEET TESTING SERVICES, INC.

Shawn K. King

DERT/DARF Consultant

SKK/ja

Enclosures

Form # 2023CL Rev. A dated 10/10/16



		12 SE	COND V	ERTICAL	BURN TE	ST				
Date:	04/26/18			AccuFleet ⁻	Test Number:	V18	3 1839			
Client:	CellBlock	FCS		Purchase C	order No.:	Dylan1				
Contact:	Dylan Van	demark		Work Orde	r No.:	18-4160				
AIRCRAFT I	NFORMAT	ION:	A/C Make:	N/A	A/C Model:	N/A	A/C S/N: N/A			
Application of M				IAL DESCR	IPTION					
Type: LIBIK out: Part Number: 3/ Color: Blue				×						
			TE	STING DAT	Ά					
Conditioning	Date:	04/24/18		Conditionin	g Time:	9:50 AM				
Date Tested:		04/26/18		Time Teste	d:	2:42 PM				
			TE	ST RESUL	rs					
	BURN	CHAR	DRIP		R	EMARKS				
-	TIME	LENGTH	TIME							
WARP	N/A	N/A	N/A							
FILL	N/A	N/A	N/A							
1	0.4	3.5	0.0							
2	0.4	4.4	0.0							
3	0.3	4.0	0.0							
AVERAGE	0.4	4.0	0.0	TEST RESULTS: PASSED						

Materials were tested in accordance with Title 14 CFR, PART 25—Subpart D, § 25.853 (a) compartment interiors, [Amdt. 25-116, Nov. 26, 2004] Appendix F to Part 25, Part I (a)(1)(ii) [Amdt. 25-142, Eff. 4/18/2016] Flame is applied for 12 seconds. The test specimen must be self-extinguishing. The average burn length must not exceed 8 inches; the average after-flame time must not exceed 15 seconds; and flaming drippings must not continue to burn for more than 5 seconds after falling to the cabinet floor.

This test was conducted in accordance with the standards established by The Federal Aviation Administration, as described above. The material is not tested by any other standard. This test is limited only to the material herein above described and delivered for testing.

The content of this data does not constitute FAA approval or all data necessarily required for the entire alteration/repair concerning installation on the aircraft. This data only demonstrated compliance with Title 14 CFR, PART 25—Subpart D, § 25.853 (a) compartment interiors, [Amdt. 25-116, Nov. 26, 2004] Appendix F to Part 25, Part I (a)(1)(ii) [Amdt. 25-142, Eff. 4/18/2016].

WITNESSED AND APPROVED BY:

Shawn K. King, DERT Consultant Form # 2023Certs Rev. A dated 10/10/16



		12 SE	COND V	ERTICAL	BURN TE	ST			
Date:	04/26/18			AccuFleet Test Number: V		V18	3 1840		
Client:				Purchase O	rder No.:	Dylan1			
Contact:	Dylan Van	demark		Work Order No.:		18-4160			
AIRCRAFT I	NFORMAT	ION:	A/C Make:	N/A	A/C Model:	N/A	A/C S/N: N/A		
				IAL DESCRI	PTION				
Application of M		Aircraft: LIBIK	Bag/Blanket		<u>, </u>				
Type: LIBIK insi Part Number: P									
Color: Black	GL2000								
Color, black									
			TE	STING DAT	A				
Conditioning	Date:	04/24/18		Conditioning	g Time:	9:50 AM			
Date Tested:		04/26/18		Time Tested	d:	2:47 PM			
			TE	ST RESULT	S				
	BURN	CHAR	DRIP		R	EMARKS			
	TIME	LENGTH	TIME						
WARP	N/A	N/A	N/A						
FILL	N/A	N/A	N/A	1					
1	0.3	0.0	0.0	1					
2	0.4	0.0	0.0						
3	0.3	0.0	0.0						
AVERAGE	0.3	0.0	0.0	TEST RESULTS: PASSED					

Materials were tested in accordance with Title 14 CFR, PART 25—Subpart D, § 25.853 (a) compartment interiors, [Amdt. 25-116, Nov. 26, 2004] Appendix F to Part 25, Part I (a)(1)(ii) [Amdt. 25-142, Eff. 4/18/2016] Flame is applied for 12 seconds. The test specimen must be self-extinguishing. The average burn length must not exceed 8 inches; the average after-flame time must not exceed 15 seconds; and flaming drippings must not continue to burn for more than 5 seconds after falling to the cabinet floor.

This test was conducted in accordance with the standards established by The Federal Aviation Administration, as described above. The material is not tested by any other standard. This test is limited only to the material herein above described and delivered for testing.

The content of this data does not constitute FAA approval or all data necessarily required for the entire alteration/repair concerning installation on the aircraft. This data only demonstrated compliance with Title 14 CFR, PART 25—Subpart D, § 25.853 (a) compartment interiors, [Amdt. 25-116, Nov. 26, 2004] Appendix F to Part 25, Part I (a)(1)(ii) [Amdt. 25-142, Eff. 4/18/2016].

WITNESSED AND APPROVED BY:

Shawn K. King, DERT Consultant Form # 2023Certs Rev. A dated 10/10/16



		12 SE	COND V	ERTICAL	BURN TE	ST			
Date:	04/26/18	4/26/18 AccuFleet Test Number: V18 1							
Client:	CellBlock	FCS		Purchase Or	rder No.:	Dylan1			
Contact:	Dylan Van	demark		Work Order	No.:	18-4160			
AIRCRAFT II	NFORMAT	ION:	A/C Make:	N/A	A/C Model:	N/A	A/C S/N: N/A		
	production								
				IAL DESCRII	PTION				
Type: LIBIK Sec Part Number: So Color: Grey	Application of Material in the Aircraft: LIBIK Bag Type: LIBIK Secondary Layer Part Number: SGL1700 Color: Grey TESTING DATA								
Conditioning		04/24/18		Conditioning	Time:	9:50 AM			
Date Tested:		04/26/18	943 365,23 9550	Time Tested		3:00 PM			
			TE	ST RESULT	S				
	BURN	CHAR	DRIP		R	EMARKS			
	TIME	LENGTH	TIME						
WARP	N/A	N/A	N/A						
FILL	N/A	N/A	N/A						
1	0.3	0.4	0.0						
2	0.4	0.4	0.0						
3	3 0.3 0.0								
AVERAGE	0.3	0.4	0.0	TEST RESULTS: PASSED					

Materials were tested in accordance with Title 14 CFR, PART 25—Subpart D, § 25.853 (a) compartment interiors, [Amdt. 25-116, Nov. 26, 2004] Appendix F to Part 25, Part I (a)(1)(ii) [Amdt. 25-142, Eff. 4/18/2016] Flame is applied for 12 seconds. The test specimen must be self-extinguishing. The average burn length must not exceed 8 inches; the average after-flame time must not exceed 15 seconds; and flaming drippings must not continue to burn for more than 5 seconds after falling to the cabinet floor.

This test was conducted in accordance with the standards established by The Federal Aviation Administration, as described above. The material is not tested by any other standard. This test is limited only to the material herein above described and delivered for testing.

The content of this data does not constitute FAA approval or all data necessarily required for the entire alteration/repair concerning installation on the aircraft. This data only demonstrated compliance with Title 14 CFR, PART 25—Subpart D, § 25.853 (a) compartment interiors, [Amdt. 25-116, Nov. 26, 2004] Appendix F to Part 25, Part I (a)(1)(ii) [Amdt. 25-142, Eff. 4/18/2016].

WITNESSED AND APPROVED BY:

Shawn K. King, DERT Consultant Form # 2023Certs Rev. A dated 10/10/16



	- 1900	12 SF	CONDV	ERTICAL	BURN TE	ST			
Date:									
Client:	CellBlock	FCS				Dylan1	5 1042		
Contact:	Dylan Van					18-4160			
AIRCRAFT II			A/C Make:				A/C S/N: N/A		
	THE OTHER	1011.	17 VO Marc.	14/7 (-vo Model.	11//	NO 0/14. 14/A		
			MATER	IAL DESCRII	PTION				
Type: LIBIK Insu Part Number: At Color: White	Application of Material in the Aircraft: LIBIK Bag Type: LIBIK Insulation Part Number: ASM250 Color: White TESTING DATA Conditioning Date: 04/24/18 Conditioning Time: 9:50 AM								
			TE	ST RESULT	S				
	BURN TIME	CHAR LENGTH	DRIP TIME	REMARKS					
WARP	N/A	N/A	N/A						
FILL	N/A	N/A	N/A						
1	0.3	0.0	0.0						
2	0.3	0.0	0.0						
3	0.4	0.0	0.0						
AVERAGE	0.3	0.0	0.0	TEST RESULTS: PASSED					

Materials were tested in accordance with Title 14 CFR, PART 25—Subpart D, § 25.853 (a) compartment interiors, [Amdt. 25-116, Nov. 26, 2004] Appendix F to Part 25, Part I (a)(1)(ii) [Amdt. 25-142, Eff. 4/18/2016] Flame is applied for 12 seconds. The test specimen must be self-extinguishing. The average burn length must not exceed 8 inches; the average after-flame time must not exceed 15 seconds; and flaming drippings must not continue to burn for more than 5 seconds after falling to the cabinet floor.

This test was conducted in accordance with the standards established by The Federal Aviation Administration, as described above. The material is not tested by any other standard. This test is limited only to the material herein above described and delivered for testing.

The content of this data does not constitute FAA approval or all data necessarily required for the entire alteration/repair concerning installation on the aircraft. This data only demonstrated compliance with Title 14 CFR, PART 25—Subpart D, § 25.853 (a) compartment interiors, [Amdt. 25-116, Nov. 26, 2004] Appendix F to Part 25, Part I (a)(1)(ii) [Amdt. 25-142, Eff. 4/18/2016].

WITNESSED AND APPROVED BY:

Shawn K. King, DERT Consultant Form # 2023 Certs Rev. A dated 10/10/16



April 26, 2018

Dylan Vandemark CellBlock FCS, LLC 234 Northeast Road Unit #5 Standish, Maine 04084

Dear Mr. Vandemark:

Enclosed are the results of the 12 Second Vertical Burn Test for CellBlock FCS, LLC Standish, ME. Test specimen description as follows:

TEST	TEST SAMPLE PART
NUMBER(S)	NUMBER(S)
V18 1843	3A, SGL1700,ASM250,PGL2000

The tests were conducted in accordance with Aircraft Materials Fire Test Handbook Chapter 1 dated 02/13/2018 to comply with Title 14 CFR, PART 25—Subpart D, § 25.853 (a) compartment interiors, [Amdt. 25-116, Nov. 26, 2004] Appendix F to Part 25, Part I (a)(1)(ii) [Amdt. 25-142, Eff. 4/18/2016], and FAA Approved "AccuFleet Testing Procedures" dated May 1995. The test operator was Maria Reyes and witness present was Shawn K. King, DERT Consultant.

If you have any questions regarding this test, please call me at 281-999-8800 extension 1111 or 1134.

Sincerely.

ACQUFLEET TESTING SERVICES, INC.

Shawn K. King

DERT/DARF Consultant

SKK/ja

Enclosures

Form # 2123CLT Rev A dated 3/15/18



		IO OFOOND W	/EDTION							
12 SECOND VERTICAL BURN TEST										
Date:	04/26/18		V18	1843						
Client:	CellBlock FCS		Purchase Order No.: Dylan1							
Contact:	Dylan Vandem	ark	Work Orde	r No.:	18-4160					
AIRCRAFT INFORMATION										
A/C Make:	A/C Make: N/A A/C Model: N/A A/C									
TEST SAMPLE PART NUMBER: 3A, SGL1700,ASM250,PGL2000 MATERIAL DESCRIPTION										
Application of Ma	aterial in the Aircraf	IVIA I EKI	IAL DESCR	IPTION						
Material #1: Type	e: LIBIK outside: Pa	irt Number: 3A; Color:	Blue							
Material #2: Type	e: LIBIK Secondary	Layer; Part Number:	SGL1700: Cold	or: Grev						
Material #3: Type	e: LIBIK Insulation;	Part Number: ASM250	0; Color: White)						
Material #4: Type	e: LIBIK inside; Parl	Number: PGL2000; 0	Color: Black							
Γ -										
Pack Side of Spe	on consists of Finish	for Material # 1 throu	igh Material #4							
Dack Side of Spe	ecimen consists of i	Material #4 through Fi	nish for Materi	al #1						
		CONDITIO	NING INFO	DMATION						
Date In:	04/24/18	CONDITIO	Time In:	9:50 AM						
Date Out:	04/26/18	Time III. 0.00 7 W								
	0 1.20, 10	TE	ST RESULT							
	FΔC	E OF SPECIMEN			K SIDE OF SPEC	-IMEN				
		CHAR LENGTH		BURN	CHAR LENGTH					
	BOILT IIIIL	OTAK ELIKOTTI	TIME	TIME	CHAR LENGTH	DRIP IIIVIE				
WARP	N/A	N/A	N/A	N/A	N/A	N/A				
FILL	N/A	N/A	N/A	N/A	N/A N/A	N/A				
1	0.4	0.1	0.0	0.3	0.0					
2	0.4	0.1	0.0	0.3		0.0				
3	0.3	0.2	0.0	0.4	0.0	0.0				
AVERAGE	0.3	05000 A			0.0	0.0				
		0.1	0.0	0.3	0.0	0.0				
RESULT:	PASSED				5					

Materials were tested in accordance with Aircraft Materials Fire Test Handbook Chapter 1 (1.6.2 Test Procedure, 1.6.2.4) dated 02/13/2018 to comply with Title 14 CFR, PART 25—Subpart D, § 25.853 (a) compartment interiors, [Amdt. 25-116, Nov. 26, 2004] Appendix F to Part 25, Part I (a)(1)(ii) [Amdt. 25-142, Eff. 4/18/2016] Flame is applied for 12 seconds. The test specimen must be self-extinguishing. The average burn length must not exceed 8 inches; the average after-flame time must not exceed 15 seconds; and flaming drippings must not continue to burn for more than 5 seconds after falling to the cabinet floor.

This test was conducted in accordance with the standards established by The Federal Aviation Administration, as described above. The material is not tested by any other standard. This test is limited only to the material herein above described and delivered for testing.

The content of this data does not constitute FAA approval or all data necessarily required for the entire alteration/repair concerning installation on the aircraft. This data only demonstrated compliance with Aircraft Materials Fire Test Handbook Chapter 1 dated 02/13/2018 and Title 14 CFR, PART 25—Subpart D, § 25,853 (a) compartment interiors, [Amdt. 25-116, Nov. 26, 2004] Appendix F to Part 25, Part I (a)(1)(ii) [Amdt. 25-142, Eff. 4/18/2016].

WITNESSED AND APPROVED BY:

Shawn K. King, DERT Consultant Form # 2123TCerts Rev A dated 3/15/18